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ParMooN – a modernized program package based on mapped finite elements

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Abstract

PARMOON is a program package for the numerical solution of elliptic and parabolic partial differential equations. It inherits the distinct features of its predecessor MOONMD [28]: strict decoupling of geometry and finite element spaces, implementation of mapped finite elements as their definition can be found in textbooks, and a geometric multigrid preconditioner with the option to use different finite element spaces on different levels of the multigrid hierarchy. After having presented some thoughts about in-house research codes, this paper focuses on aspects of the parallelization, which is the main novelty of PARMOON. Numerical studies, performed on compute servers, assess the efficiency of the parallelized geometric multigrid preconditioner in comparison with parallel solvers that are available in external libraries. The results of these studies give a first indication whether the cumbersome implementation of the parallelized geometric multigrid method was worthwhile or not.

1. Some Considerations about In-House Research Codes

MOONMD, a C++ program package for the numerical solution of elliptic and parabolic partial differential equations based on mapped finite elements, has been described in [28]. A modernized version of this package, called PARMOON, has been recently developed to be used as a research code in the future.

Nowadays, several academic software packages for solving partial differential equations exist in the research community. They are usually developed and supported by research groups for whom software development is one of the main scientific tasks. Such software packages include, among others, DEAL.II [10], FENICS [2], DUNE [15, 11], or OPENFOAM [32]. These packages have advanced functionality and support features like adaptive mesh refinement, parallelism, etc.

Naturally, a research code developed in house possesses less functionality than these large packages. In view of their availability, the following questions arise: Why is it worth to develop an own research code? In particular, is it worth to develop a code within a research group that focuses primarily on numerical analysis? In the following, some arguments, mainly based on our own experience, are presented.

The first key aspect of working with a code developed and maintained within the research group is the detailed knowledge of the software structure. In fact, applied mathematicians often work at the development of algorithms. These algorithms have to be implemented, assessed and compared with popular stateof-the-art algorithms for the same problem. A meaningful assessment requires the usage of the algorithms in the same code. In this respect, it is important to have access to a code where one knows and can control every detail.

For brevity, just one example will be mentioned that shows the importance of knowing the details of a software package. This example concerns the clarification of appropriate interface conditions in subdomain iterations for the Stokes– Darcy problem, see [14]. Standard Neumann interface conditions can be used only for viscosity and permeability coefficients that are unrealistically large. For realistic coefficients, appropriate Robin boundary conditions have to be used. The implementation of the Robin interface conditions was performed in a straightforward way in MOONMD/PARMOON.

Further advantages of an own research code are the possibility of controlling its core parts and flexibility. In particular, for our research it is very important that the code supports the use of different discretization strategies. As an example, MOONMD was designed for finite element methods. But for the investigation of discretizations for time-dependent convection-diffusion equations in [29], finite difference methods were implemented as well. Because these methods performed very well, they were later used in the context of simulating population balance systems defined in tensor-product domains, e.g., see [12, 36].

The list of examples could be extended. In addition, a number of numerical algorithms have been developed and implemented in our research code which turned out to be not (yet) competitive, like the optimization of stabilization parameters in SUPG methods in [27]. Having a known and flexible research code at disposal allows to test and support algorithms that, at the time of the implementation, have not been benchmarked in detail.

Certainly, also the large packages mentioned above allow the implementation of different algorithms and discretization strategies. However, we think that a successful implementation often requires a very close interaction with core developers of the packages. This effort might not be feasible for both the user and the developer. Therefore, an own code might reduce the time from the development to the assessment of numerical methods.

A further aspect, related with the interaction of the core development team, concerns the students and the PhD students who are involved in the development and in the usage of the code. Since the core developers of an own research code are readily available and they are experts in the focused research topics of the group, these students can be supported efficiently. In addition, students working at the code stated several positive effects: the work at details of the implementation facilitates the insight into the methods, which is important for analyzing their properties, and the skills in software design and management are enhanced. Of course, incorporating students into code development requires that there is an easy use of the code and an easy access to basic routines, such as, in the case of a finite element solver, assembling of matrices and solving linear systems of equations.

This issue touches already the next question: What should be expected from an own research code?

In order to support students when starting to work with the code, an easy installation and basic testing setup are essential. There are even successful attempts for designing complex codes that can be used for teaching students in basic courses on numerical methods for partial differential equations, like the so-called computational laboratory for Investigating Incompressible Flow Problems (IFISS), see [16], which uses MATLAB, and the open source software FREEFEM++ [19], which is based on an own language.

The code should be modular. In particular, there should be a general core and individual projects should be attached to this core. Of course, the projects use routines from the core. But with an own code, it seems to be easier than with a large package, which is developed somewhere else, to incorporate contributions from the projects into the core.

With respect to the required stability of the code, there are, in our opinion, no fundamental differences between own research codes and large packages.

However, there are different expectations with respect to the efficiency. A research code should be flexible in many respects, since its main tasks include supporting the development of numerical methods and results from numeri-

cal analysis. For instance, in the code MOONMD/PARMOON, the concept of mapped finite elements is implemented, see Section 3. In this way, the code supports currently around 170 finite elements in two dimensions and 75 finite elements in three dimensions. Consequently, all routines are implemented for the general situation. For certain finite elements, this might be less efficient than using tailored routines. However, also for a research code, efficiency is a key property that should not be neglected. For instance, the simulation of standard academic benchmark problems for turbulent incompressible flows requires the computation of large time intervals to collect temporal averages of statistics of interest. In our opinion, an own research code should be reasonably efficient on the available in-house hardware, which, in our case, are usually laptops, compute servers, or small clusters.

2. On the Objectives of this Paper

One of the aims of this paper is to report on the development of the existing research code MOONMD towards a new package, in order to accomplish the goals mentioned in Section 1. The original code possesses some distinct features, like the strict decoupling of geometry and finite element spaces, the implementation of mapped finite elements as their definition can be found in textbooks, or a multiple discretization multilevel (MDML) preconditioner, see Section 5. The core of MOONMD has been designed more than 15 years ago and this code has been successfully used so far in many scientific studies. There are almost 90 research articles citing MOONMD via [28], see [37]. Recent advances in computing hardware and language standards necessitate a re-design and re-implementation of some of the core routines. With the new core and the new features, the code was renamed to PARMOON (Parallel Mathematics and object-oriented Numerics).

This paper focuses on the most relevant aspect concerning the development of PARMOON, namely the parallelization. In particular, the technically most cumbersome part, the parallelization of the geometric multigrid, is discussed.

Its main contribution is a first assessment of the resulting parallel geometric multigrid method in comparison with parallel solvers for linear systems of equations that can be called from external libraries. The numerical studies were performed on compute servers, which is the available in-house hardware. We think that this assessment is also of interest for other groups who develop their own code in order to get an impression whether it is worthwhile to implement a parallelized geometric multigrid method or not. Two main problem classes supported in PARMOON are considered in the numerical studies: scalar convection-diffusion-reaction equations and the incompressible Navier–Stokes equations.

3. Mapped Finite Element Spaces

The implementation of finite element methods in MOONMD/PARMOON is based on a rather abstract definition of a finite element space and on the

mapping of each mesh cell to a reference mesh cell.

Let $\Omega \subset \mathbb{R}^d$, $d \in \{2, 3\}$, be a bounded domain and let \mathcal{T}^h be some triangulation of Ω consisting of compact mesh cells. For each mesh cell K, a local finite element space $P(K) \subset C^s(K)$, $s \geq 0$, is given by some finite-dimensional space of functions spanned by a basis $\{\phi_{K,i}\}_{i=1}^{N_K}$. Furthermore, a set of local linear functionals $\{\Phi_{K,i}\}_{i=1}^{N_K}$ is given. The space P(K) is unisolvent with respect to the functionals. Often, a so-called local basis is chosen, i.e., a basis that satisfies $\Phi_{K,i}(\phi_{K,j}) = \delta_{ij}$ for $i, j = 1, \ldots, N_K$. The local linear functionals might be values of the functions or their derivatives in certain points, integrals on K or on faces of K.

Let $\Phi_1, \ldots, \Phi_N : C^s(\overline{\Omega}) \to \mathbb{R}$ be given continuous linear functionals. The restriction of each functional to $C^s(K)$ defines the set of local functionals. The union of all mesh cells K_j , for which there is a $p \in P(K_j)$ with $\Phi_i(p) \neq 0$, will be denoted by ω_i . Now, the global finite element space is defined as follows. A function $v(\boldsymbol{x})$ defined on Ω with $v|_K \in P(K)$ for all $K \in \mathcal{T}^h$ is called continuous with respect to the functional $\Phi_i : C^s(\overline{\Omega}) \to \mathbb{R}$ if $\Phi_i(v|_{K_1}) = \Phi_i(v|_{K_2})$ for all $K_1, K_2 \in \omega_i$. The space

$$S = \left\{ v \in L^{\infty}(\Omega) : v|_{K} \in P(K) \text{ and } v \text{ is continuous with respect to} \right.$$
$$\Phi_{i}, i = 1, \dots, N \right\}$$

is called finite element space. The global basis $\{\phi_j\}_{j=1}^N$ of S is defined by the conditions $\phi_j \in S$ with $\Phi_i(\phi_j) = \delta_{ij}$ for $i, j = 1, \ldots, N$.

- Using this definition for the implementation of a finite element space requires 1.) the definition of the local basis and linear functionals for each K,
- 2.) the implementation of a method that assures continuity stated in the definition of S.

The first requirement can be achieved in two different ways, via a mapped or an unmapped implementation. In the unmapped approach, the local basis and linear functionals are defined directly on K. In contrast, mapped finite elements are closely connected to a standard way of analyzing finite element discretizations. This analysis consists of three steps:

- Map an arbitrary mesh cell K to a compact reference mesh cell \hat{K} .
- Prove the desired properties on \hat{K} , which is the core of the analysis.
- Map the reference mesh cell \hat{K} back to K to get the final result.

Hence, this approach has two main features:

- All considerations have to be done on \hat{K} only.
- There is no information necessary and available about neighbor mesh cells of *K*.

Mapped and unmapped finite element methods possess the same analytical properties if the reference map $F_K : \hat{K} \to K$ is affine for every mesh cell K of the given triangulation. In the case of non-affine maps, occurring, e.g., for a triangulation consisting of arbitrary quadrilateral or hexahedral mesh cells, mapped and unmapped finite element spaces might be different. In MOONMD/

PARMOON, the concept of mapped finite elements is implemented in the following way. Reference mesh cells are the unit simplices, e.g., in two dimensions with the vertices (0,0), (1,0), (0,1), and the unit cubes $\hat{K} = [-1,1]^d$. Affine maps are available for all reference mesh cells. To account for arbitrary quadrilaterals and hexahedra, *d*-linear maps are also implemented for the unit cubes. Based on the different reference cells, local spaces on \hat{K} , linear functionals, and reference maps, MOONMD/PARMOON currently supports about 170 finite elements in two dimension and 75 finite elements in three dimensions.

The use of mapped finite element spaces essentially requires the implementation of finite elements on the reference cells. The quadrature rules for numerical integration have to be implemented only on these cells, since the integrals on physical cells are transformed to integrals on the reference cells. Note that the same strategy works also for the handling of cell faces, which are mapped onto lower-dimensional reference cells by corresponding reference maps.

A d.o.f.-manager (d.o.f. - degree of freedom) is used in MOONMD/PARMOON for ensuring the continuity in the definition of the finite element space. For a detailed description of the d.o.f.-manager, see [28].

The implementation of the concepts described in this section have been adapted from MOONMD to PARMOON in a straightforward way. It is clear that the definition of the local basis and functionals is completely local and therefore not affected by the parallelization. The required continuity in the definition of finite element spaces is performed in PARMOON on each process separately. The computed results are visualized with the software package PARAVIEW [1, 6] that does not require global numbers for the d.o.f.s. However, if needed, global numbers for d.o.f.s can be assigned, compare [18]. Altogether, the used concepts turned out to be applicable in the same way as well for the sequential as for the parallel code.

4. Parallel Data Structures in PARMOON

PARMOON supports a single program, multiple data (SPMD) approach on parallelism, making use of the Message Passing Interface (MPI) standard [31]. It relies on a decomposition of the domain, which is the standard for parallelized finite element codes. Decomposing the computational domain and distributing it among the processes naturally leads to a parallelization of matrix-vector operations. The local aspect of the finite element method, which is reflected in the sparsity of the arising matrices, limits the communication overhead.

4.1. Decomposing the Domain - Own Cells and Halo Cells

In order to distribute the domain among the participating MPI processes, where each process executes a MPI parallel program, PARMOON makes use of the METIS graph partitioning tool [30]. At program start, all processes read the same geometry and perform the same initial domain refinement steps. Upon reaching the first refinement level on which computations will be performed, the root process (process number 0) calls the METIS library to compute a disjoint

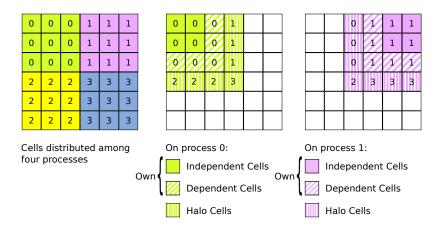


Figure 1: Different cell types due to a domain decomposition.

domain decomposition, i.e., to determine which process is going to be in charge of which mesh cells.

Root then communicates the METIS output to the other processes. Each process is informed about the cells it will be responsible for. These cells are called *own cells* of the process. Each process then maintains only its own cells plus those cells that share a boundary face, edge, or vertex with an own cell. In domain decomposition methods these cells are commonly referred to as *halo cells*. The sketches in Figure 1 clarify that expression – the halo cells form a one-layer thick halo around the set of own cells.

The own cells are further divided into *dependent* and *independent cells*, denoting with *interface* the set of faces shared by own cells and halo cells. All own cells which share a piece of interface are called dependent cells, while the remaining own cells are called independent cells.

The requirements on an efficient domain decomposition are twofold: the computational load must be balanced, i.e., there should be a comparable number of cells on each process, and the needed amount of communication must be small, i.e., the interfaces should be as small as possible. Due to the deletion of cells, each process stores only a part of the entire problem (multiple data), but all processes execute the same program code (single program). With its domain reduced to own cells plus halo cells, each process sets up a finite element space and perform all further computations only on the own part of the domain.

4.2. Types of Degrees of Freedom

A d.o.f. with index *i* is defined by the finite element basis function ϕ_i and the associated global linear functional Φ_i . It is represented, e.g., by the *i*-th entry in the vectors for the solution and the right-hand side.

A d.o.f. on a certain process P will be classified depending on the localization of the associated finite element basis function and on the classes of d.o.f.s it is coupled with. Two degrees of freedom of a finite element in d dimensions are said to be *coupled* if the supports of the corresponding basis functions intersect on a set of non-zero *d*-dimensional measure. Note that, with this definition, coupling only occurs for d.o.f.s that are located in the same mesh cell. The coupling of two d.o.f.s transfers directly to a property of the finite element matrix A: the coupling of d.o.f. *i* and *j* will lead potentially to non-zero entries a_{ij} and a_{ji} and the needed memory for these entries has to be allocated.

In the following, the classes for handling d.o.f.s in PARMOON will be described shortly. All d.o.f.s that are localized in a cell known to processor P are called *known* d.o.f.s and will be denoted by D_{known}^P (D_*^P stands in the following for number of d.o.f.s on P of type *). This set of d.o.f.s is then divided into *master* and *slave* d.o.f.s, i.e.,

$$D_{\mathrm{known}}^P = D_{\mathrm{master}}^P \stackrel{\cdot}{\cup} D_{\mathrm{slave}}^P$$

where \cup denotes a disjoint union. A d.o.f. *i* is said to be a master d.o.f. on P if P is responsible for the value of *i*, in a way that will be clarified below. All known d.o.f.s which are not master on P are called *slave* d.o.f.s. It is worth noticing that every d.o.f. in the entire problem is a master on exactly one process. However, a d.o.f. can be slave on more than one process.

One can introduce the following classes of d.o.f.s:

- Independent d.o.f., i.e., all d.o.f.s which lie in P's own cells but not in its dependent cells. All P's independent d.o.f.s are set as master d.o.f.s, since they are not even known to any other process. They only couple to other master d.o.f.s of P.
- Dependent d.o.f., i.e., those d.o.f.s lying in P's dependent cells, but not in its halo cells. Process P is the master of all its dependent d.o.f.s. The notation is motivated by the fact that the dependent d.o.f.s are in a vicinity to the domain interface and therefore possess, to some extent, a dependency on other processes.
- Interface d.o.f., i.e., all d.o.f.s that lie on the intersection of dependent cells and halo cells. These d.o.f.s are known to all adjacent processes as interface d.o.f., too. Only one of these processes will take master responsibility for each interface d.o.f. In particular, on a process *P*, one distinguishes between master interface d.o.f. (the interface d.o.f.s which are master on *P*) and slave interface d.o.f. (all others d.o.f.s, for which neighboring processors take master responsibility).
- *Halo d.o.f.*, i.e., all d.o.f.s which lie in halo cells but not on the interface. Since all of them are dependent d.o.f.s to neighboring processes, one of these will take master responsibility for them. On *P* all halo d.o.f.s are slave d.o.f.s.

Hence, the sets of master and slave d.o.f.s can be divided as

$$\begin{split} D^P_{\text{master}} &= D^P_{\text{independent}} \stackrel{\cdot}{\cup} D^P_{\text{dependent}} \stackrel{\cdot}{\cup} D^P_{\text{interface master}},\\ D^P_{\text{slave}} &= D^P_{\text{interface slave}} \stackrel{\cdot}{\cup} D^P_{\text{halo}}. \end{split}$$

In general, the d.o.f. classification depends on the finite element spaces and the

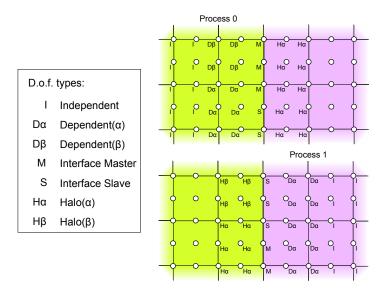


Figure 2: Types of d.o.f.s at the interface for two-dimensional Q_2 finite elements, from the point of view of process 0 and process 1.

decomposition of the domain. Figure 2 sketches the different types of d.o.f.s for two-dimensional Q_2 finite elements at the interface between two processes.

Furthermore, it is convenient to further refine the d.o.f. classification in order to reduce the communication overhead, see also Section 4.3. To this aim, the halo and the dependent d.o.f.s of P are further divided into $\text{Halo}(\alpha)$ and $\text{Halo}(\beta)$, and $\text{Dependent}(\alpha)$ and $\text{Dependent}(\beta)$, respectively:

- $Halo(\alpha)$ and $Halo(\beta)$ d.o.f.s.: The Halo(α) d.o.f.s are those that are coupled with at least one (interface) master d.o.f. of P, while Halo(β) d.o.f.s are coupled solely with other slave d.o.f.s, i.e., with interface slave d.o.f.s and other halo d.o.f.s.
- Dependent(α) and Dependent(β) d.o.f.s.: Dependent(α) d.o.f.s are those connected to at least one (interface) slave d.o.f., while Dependent(β) d.o.f.s are all those that are connected to master d.o.f.s only, i.e., to interface master, other dependent, or independent d.o.f.s. Note that all Dependent(β) d.o.f.s of process P will be Halo(β) d.o.f. on all other processes where they are known. For Dependent(α) d.o.f.s the situation is not as simple. Each of them is Halo(α) to at least one neighboring process, but can be Halo(β) to others, see also the sketch in Figure 2.

4.3. Consistency Levels

In parallel computations, values can be stored either in a consistent way or in an inconsistent (or additive) way. *Consistent* storage means that all processes have the same and correct value at all respective d.o.f.s as in the sequential environment. In *additive* storage, a global value is the sum of the values over all processes where it is known. In PARMOON mainly consistent storage and weakened concepts hereof are used.

Four stages of consistency are important for PARMOON finite element objects, like vector representations of a finite element function or a matrix.

- Level-0-consistent. Consistency holds only with regard to master d.o.f.s. Each master d.o.f. on each process holds the same value as it would in a sequential computation. The values of slave d.o.f.s are in an undefined storage state. In the implementation of operations care must be taken of not losing Level-0-consistency all master values must be kept as they would be in a sequential execution.
- Level-1-consistent. All master d.o.f.s and all interface slave d.o.f.s are consistent. The values of all halo d.o.f.s are in an undefined storage state.
- Level-2-consistent. Consistency is established for all but $\text{Halo}(\beta)$ d.o.f.s. The values of $\text{Halo}(\beta)$ d.o.f.s. are in an undefined storage state while all other values are consistent.
- *Level-3-consistent*. All d.o.f.s are stored consistently. This is the "real" consistent storage.

The main motivations behind introducing this classification are that different operations require a different state of consistency of their input data and that restoring a certain state of consistency requires a certain amount of communication – the lower the required state of consistency, the lower the required amount of communication.

After the domain was decomposed, each process P assembles a finite element matrix on all its known cells. The use of halo cells assures that all information to assemble the rows belonging to master d.o.f.s is available on P. The finite element matrix assembled in this way will therefore be row-wise Level-1-consistent – all rows belonging to master and slave interface d.o.f. are correctly assembled.

Multiplication of a row-wise Level-0-consistent matrix with a Level-2-consistent vector gives a Level-0-consistent vector. Level-3-consistency of the vector factor is not needed. If the vector factor is Level-3-consistent, together with the matrix being Level-1-consistent, the result will be even Level-1-consistent. There is no further benefit in keeping the matrix in higher consistency levels than Level-0-consistency.

Multiplying a vector with a constant scalar will maintain the current consistency level, as will vector-vector addition. In the latter case, vectors with different consistency level might be added and the result has the lower consistency level. Scalar products require Level-0-consistency of both vectors, where all slave d.o.f.s will be skipped, and a globally additive reduce operation to get a consistent result.

Level-3-consistency of a finite element vector in PARMOON is only enforced if operations require knowledge of the represented finite element function even on the halo cells. Such operations are for example matrix assembling with an input finite element function, like for the convective term of the Navier–Stokes equations, grid transfer operations in multigrid methods, or gradient recovery by averaging gradients over a patch of mesh cells.

Relation (shorthand)	Master type	updates	Slave type
Interface (IMS)	Interface Master	\longrightarrow	Interface Slave
$Dependent(\alpha) - Halo(\alpha) (DH\alpha)$	$Dependent(\alpha)$	\longrightarrow	$\operatorname{Halo}(\alpha)$
$Dependent(\beta) - Halo(\beta) (DH\beta)$	$\operatorname{Dependent}(\beta)$	\longrightarrow	$\operatorname{Halo}(\beta)$

Table 1: Master-slave relationship of d.o.f. types.

Enforcing certain consistency levels is a matter of communication. For each d.o.f. which needs an update, the responsible master process communicates its value to all processes where it is slave. These processes simply reset its value to the received value. The required infrastructure for this communication is set up just once for a certain finite element space and can be reused whenever an update is necessary.

4.4. Organizing Communication

When setting up the communication structure, one has to find, for each non-independent master d.o.f. i, all those slave d.o.f.s on other processes that are globally identical to i. Certain master types match with certain slave types, see Table 1, forming three distinct pairs of master-slave relations. To restore a certain consistency level, at least one of these relations requires an update.

Note that it is not possible in general to immediately identify the global number of a d.o.f., since each process creates finite element spaces only on its own cells and it numbers its d.o.f.s locally, unaware of the other processes. In PARMOON, the global identification of a d.o.f. is defined according to the global number of the mesh cell in which the d.o.f. is located. These global cell numbers are assigned to each cell before decomposing the domain. Performing uniform refinements, such a globally unique cell number can easily be given to children cells, too. Hence, the globally consistent cell number and a globally consistent local d.o.f. number (within each cell) make it possible to identify each d.o.f. globally.

The communication structure is stored in a class called ParFEMapper, while the communication itself is performed by a class named ParFECommunicator. Setting up the ParFEMapper and ParFECommunicator requires some communication itself, and a detailed description of this task is presented in [18]. In what follows, only an overview of it is given, including a short description of those data fields of ParFEMapper that are relevant when updating the d.o.f.s of a certain master-slave relation, see Table 1. Since for all three relations these data fields are corresponding, the interface (IMS) relation is presented as an example.

For the IMS update, the ParFECommunicator wraps a call to the MPI function MPI_Alltoallv, where every process may send a different set of values of the same type (MPI_DOUBLE in this case) to each other process. To control the MPI_Alltoallv call, the ParFEMapper stores the following data, where mpi_size is the total number of processes and nInterfaceSlaves is the number of interface slaves local to process *P*.

- int* sendBufIMS is the send buffer, filled with the values of all interface masters, each one possibly appearing more than once, which will then be sent to the other processes. Its total length equals the sum over all values of sendCountsIMS.
- int* sendCountsIMS is an array of size mpi_size. It lists how many values P has to send to each process.
- int* sendDisplIMS is the send displacement, an array of size mpi_size. It stores where in the array sendBufIMS the message for a certain process starts. In PARMOON there are neither overlaps nor gaps, so sendDisplIMS[i] holds the sum of sendCountsIMS[0] to sendCountsIMS[i-1].
- int* recvBufIMS is the receive buffer, which will be filled with sent values from the other processes in the communication routine. Its size equals nInterfaceSlaves.
- int* recvCountsIMS is an array of size mpi_size. It lists how many values are to be received from each process. The sum of its values is nInterfaceSlaves.
- int* recvDisplIMS is the receive displacement. Like for int* send-DisplIMS there are neither gaps nor overlaps.

Besides the data, which is needed in the immediate control of MPI_Alltoallv, the ParFEMapper contains two arrays that allow to interpret the sent and received data, by mapping between send- or receive buffer and the local d.o.f.s:

- int* sentDofIMS interprets sentDofIMS[i] = d as: The *i*-th place in sendBufIMS has to be filled with the value of the local d.o.f. *i*.
- int* rcvdDofIMS interprets rcvdDofIMS[i] = d as: The *i*-th value in recvBufIMS should update the local d.o.f. *i*.

In the same way, the communication for $DH(\alpha)$ and $DH(\beta)$ is organized.

To change a vector from Level-0-consistency to Level-1-consistency, only an IMS update is required. For restoring Level-2-consistency, additionally a $DH(\alpha)$ update is necessary while Level-3-consistency requires even a $DH(\beta)$ update on top.

5. The Parallel Geometric Multigrid Method

Geometric multigrid methods are an appealing option to be used as preconditioners in problems where the necessary hierarchy of grids can be provided. These methods have been used in the simulations performed with MOONMD in particular for three-dimensional problems and for linear saddle point problems arising in the linearization and discretization of equations modeling incompressible flow problems [25].

The components of a geometric multigrid method are the following: function prolongation, defect restriction, function restriction, smoother, and coarse grid solver. Details of the algorithms and implementation of the first three components in MOONMD/PARMOON can be found in [28] and it will be referred for brevity. It shall be only noted that multilevel methods are supported that allow different finite element spaces on different levels of the multigrid hierarchy.

In particular, the so-called multiple discretization multilevel (MDML) method for higher order discretizations can be used. This method possesses the higher order discretization on the highest multigrid level and a low order discretization on all coarser levels. The motivation for this approach is the experience that multigrid methods often work very efficiently for low order discretizations. Numerical studies of the efficiency of the MDML method can be found, e.g., in [23, 24]. The grid transfer operations are performed with a local operator, taking values only on the mesh cells of the current level, proposed in [35] that can handle different finite element spaces on different levels of the grid hierarchy, see [28, 25] for details.

The implementation and parallelization of geometric multigrid methods require a considerable amount of work. The geometric data structures need to be equipped with parent-child information and the grid transfer operations have to be implemented. In the current version of PARMOON, each process is responsible of a part of the coarsest grid and refines this part uniformly, compare Section 4.1. Consequently, all parent-child information is available on the process. More technical details on constructing the grid hierarchy can be found in [18].

As usual in parallel geometric multigrid methods, block-Jacobi smoothers are applied, where the blocks correspond to the master and interface slave degrees of freedom of a process. Within the blocks, the actual smoother, like SSOR or the Vanka smoother, is used. After each smoothing iteration, the values at the interface are updated by computing their arithmetic average. As already mentioned above, the grid transfer operators need as potential input all values that are connected to a mesh cell. Since it is sufficient to perform the grid transfer only on own cells, the input vectors for the grid transfer have to be level-1-consistent.

6. Numerical Studies

The performed numerical studies are a first step of assessing the efficiency of the parallelized geometric multigrid method in comparison with parallel solvers that can be used by linking an external library to the code. The underlying question is whether it was worthwhile to perform the complex parallelization of this method. We think that this question arises also in other groups that maintain an in-house research code.

In this paper, the numerical studies concentrate on the standard multigrid method (same number of geometric and multigrid levels, same discretization on each level) since we think that this method is of most interest for the community. It was used as preconditioner in the flexible GMRES (FGMRES) method [33]. The system on the coarsest grid was solved with a sparse direct solver. Moreover, the V-cycle was applied because of its better efficiency on parallel computers compared with the more stable F- and W-cycle (the F-cycle is our standard approach in sequential simulations). The V-cycle approaches less often coarser grids, which possess an unfavorable ratio of computational work and necessary communications, than the other cycles. The efficiency of the geometric multigrid preconditioner has been compared with the efficiency of the sparse direct solver MUMPS [3, 4] and the FGMRES method preconditioned with SSOR (for scalar problems), the BoomerAMG [21] (for scalar problems), or the LSC preconditioner [17] (for linear saddle point problems). These solvers were used as they are provided from the library PETSC, version 3.7.2, [7, 8, 9]. The restart parameter in FGMRES was set to be 50.

The numerical studies were performed on a hardware platform that can be usually found in universities and academic institutes, in our case, on compute servers HP BL460c Gen9 2xXeon, Fourteen-Core 2600MHz. We think that the performance on a hardware platform that is widely available is of interest for the community. The results will consider only the computing times for the different solvers of the linear systems of equations. We could observe some variations of these times for the same code and input parameters but different runs. To reduce the influence of these variations, all simulations were performed five times, the fastest and the slowest computing time were neglected and the average of the remaining three times is presented below.

Example 1. Steady-state convection-diffusion equation. This example is a threedimensional extension of a benchmark problem for two-dimensional convectiondiffusion equations – the so-called Hemker example [20, 5]. The domain is defined by

$$\Omega = \left\{ \{ (-3,9) \times (-3,3) \} \setminus \left\{ (x,y) : x^2 + y^2 \le 1 \right\} \right\} \times (0,6)$$

and the equation is given by

$$-\varepsilon \Delta u + \boldsymbol{b} \cdot \nabla u = 0 \quad \text{in } \Omega,$$

with $\varepsilon = 10^{-6}$ and $\mathbf{b} = (1,0,0)^T$. Dirichlet boundary conditions u = 0 were prescribed at the inlet plane $\{x = -3\}$ and u = 1 at the cylinder. At all other boundaries, homogeneous Neumann boundary conditions were imposed. This example models, e.g., the heat transport from the cylinder. The solution exhibits boundary layers at the cylinder and internal layers downwind the cylinder, see Figure 3.

It is well known that stabilized discretizations have to be employed in the presence of dominant convection. In the numerical studies, the very popular streamline-upwind Petrov–Galerkin (SUPG) method [22, 13] was used with the standard parameter choice given in [26, Eqs. (5) – (7)]. Simulations were performed with Q_1 finite elements. The initial grid is depicted in Figure 3.

Computational results are presented in Figures 4 and 5 for refinement levels 4 (1 297 375 d.o.f.s) and 5 (10 388 032 d.o.f.s). The PETSC solvers were called with the flags -ksp_type fgmres -pc_type sor and -ksp_type fgmres -pc_type hypre -pc_hypre_type boomeramg, respectively. For the geometric multigrid preconditioner, the V(2,2)-cycle was used and the overrelaxation parameter of the SSOR smoother within the block-Jacobi method was set to be $\omega = 1$. This approach is certainly not optimal since with an increasing number

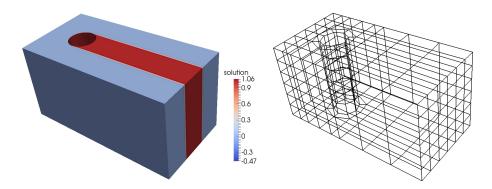


Figure 3: Example 1, numerical solution (left) and initial grid (level 0, right). The color bar shows that the numerical solution computed with the SUPG method possesses underand overshoots. These spurious oscillations occur in particular in a vicinity of the cylinder, compare [27, Fig. 14] for the two-dimensional situation.

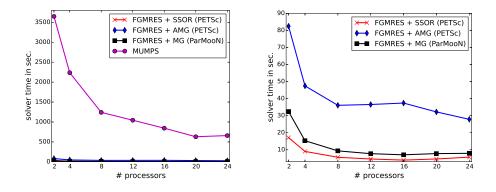


Figure 4: Example 1, solver times on refinement level 4.

of processes the number of blocks of the block-Jacobi method increases, which in turn makes it advantageous to apply some damping, i.e., a somewhat smaller overrelaxation parameter. In order not to increase the complexity of the numerical studies, we decided to fix a constant overrelaxation parameter that worked reasonably well for the whole range of processor numbers which was used. The iterative solvers were stopped if the Euclidean norm of the residual vector was less than 10^{-10} . Parameters like the stopping criterion, the overrelaxation factor, and the restart parameter were the same in all iterative methods.

It can be seen in Figure 4 that the sparse direct solver performed less efficient by around two orders of magnitude than the other solvers. This behavior was observed for all studied scalar problems and no further results with this solver for scalar problems will be presented. Among the iterative solvers, PETSC FGMRES with SSOR preconditioner and PARMOON FGMRES with geometric multigrid preconditioner (MG) performed notably more efficient than PETSC FGMRES with BoomerAMG. The latter solver did not even converge on the

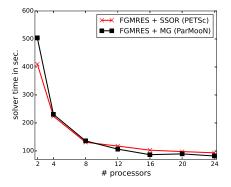


Figure 5: Example 1, solver times on refinement level 5. PETSC FGMRES with BoomerAMG converged only with two processors (1519 sec.).

finer grid in the simulations on more than two processors. FGMRES with SSOR required considerably more iterations than PARMOON FGMRES with MG: around 125 vs. 25 on level 4 and 320 vs. 45 on level 5. A notable decrease of the computing time for the iterative solvers on the coarser grid can be observed only until 8 processors. On this grid, PETSC FGMRES with SSOR was a little bit faster than PARMOON FGMRES with MG. On the finer grid, a decrease of the computing times occurred until 16 processors and PARMOON FGMRES with MG was often a little bit more efficient.

Example 2. Time-dependent convection-diffusion-reaction equation. This example can also be found in the literature, e.g., in [29], and it models a typical situation which is encountered in applications. A species enters the domain $\Omega = (0, 1)^3$ at the inlet $\Gamma_{\rm in} = \{0\} \times (5/8, 6/8) \times (5/8, 6/8)$ and it is transported through the domain to the outlet $\Gamma_{\rm out} = \{1\} \times (3/8, 4/8) \times (4/8, 5/8)$. In addition, the species is diffused somewhat and in the subregion where the species is transported, also a reaction occurs. The ratio of diffusion and convection is typical for many applications.

The underlying model is given by

$$\begin{split} \partial_t u - \varepsilon \Delta u + \boldsymbol{b} \cdot \nabla u + cu &= 0 & \text{ in } (0,3) \times \Omega, \\ u &= u_{\text{in}} & \text{ in } (0,3) \times \Gamma_{\text{in}}, \\ \varepsilon \frac{\partial u}{\partial \boldsymbol{n}} &= 0 & \text{ on } (0,3) \times \Gamma_{\text{N}}, \\ u &= 0 & \text{ on } (0,3) \times (\partial \Omega \setminus (\Gamma_{\text{N}} \cup \Gamma_{\text{in}})), \\ u(0,\cdot) &= u_0 & \text{ in } \Omega. \end{split}$$

The diffusion parameter is given by $\varepsilon = 10^{-6}$, the convection field is defined by $\boldsymbol{b} = (1, -1/4, -1/8)^T$, and the reaction by

$$c(\boldsymbol{x}) = \begin{cases} 1 & \text{if } \operatorname{dist}(\boldsymbol{x}, g) \leq 0.1, \\ 0 & \text{else}, \end{cases}$$

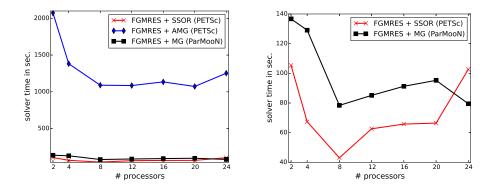


Figure 6: Example 2, solver times on the 64^3 cubed mesh.

where g is the line through the center of the inlet and the center of the outlet and dist(x, g) denotes the Euclidean distance of the point x to the line g. The boundary condition at the inlet is prescribed by

$$u_{\rm in} = \begin{cases} \sin(\pi t/2) & \text{if } t \in [0,1], \\ 1 & \text{if } t \in (1,2], \\ \sin(\pi (t-1)/2) & \text{if } t \in (2,3]. \end{cases}$$

The initial condition is set to be $u_0(\mathbf{x}) = 0$. Initially, in the time interval [0, 1], the inflow increases and the injected species is transported towards the outlet. Then, there is a constant inflow in the time interval (1, 2] and the species reaches the outlet. At t = 2, there is almost a steady-state solution. Finally, the inflow decreases in the time interval (2, 3], compare [29].

The SUPG stabilization of the Q_1 finite element method was used and as temporal discretization, the Crank–Nicolson scheme with the equidistant time step $\Delta t = 10^{-2}$ was applied. Simulations were performed on grids with 64^3 and 128^3 cubic mesh cells. The coarsest grid for the geometric multigrid method possessed 4^3 cubic mesh cells. The SSOR method was applied with the overrelaxation parameter $\omega = 1.25$. As initial guess for the iterative solvers, the solution from the previous discrete time was used. The availability of a good initial guess and the dominance of the system matrix by the mass matrix are the main differences to the steady-state case.

Computing times for the iterative solvers are presented in Figures 6 and 7. On the coarser grid, a speed-up can be observed only until 8 processors and on the finer grid until 12 to 16 processors. With respect to the efficiency of the iterative solvers, the same observations can be made as in the steady-state Example 1. The same statement holds true for the number of iterations per time step, e.g., on the finer grid there were up to 45 for PETSC FGMRES with SSOR and around 1-3 for PETSC FGMRES with BoomerAMG and PARMOON FGMRES with MG.

Example 3. Steady-state incompressible Navier-Stokes equations. This exam-

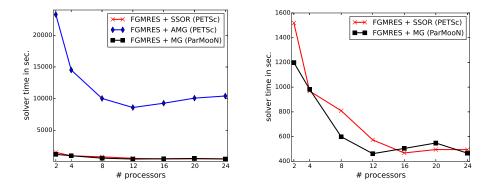


Figure 7: Example 2, solver times on the 128³ cubed mesh.

ple considers the benchmark problem of the flow around a cylinder defined in [34]. The steady-state Navier–Stokes equations are given by

$$-\nu\Delta \boldsymbol{u} + (\boldsymbol{u}\cdot\nabla)\boldsymbol{u} + \nabla p = \boldsymbol{0} \quad \text{in } \Omega,$$

$$\nabla \cdot \boldsymbol{u} = \boldsymbol{0} \quad \text{in } \Omega,$$

where \boldsymbol{u} is the velocity, p the pressure, $\nu = 10^{-3}$ is the dimensionless viscosity, and Ω is the domain given by

$$\Omega = \left\{ (0, 2.55) \times (0, 0.41) \setminus \left\{ (x - 0.5)^2 + (y - 0.2)^2 \le 0.05^2 \right\} \right\} \times (0, 0.41).$$

At the inlet x = 0, the velocity was prescribed by

$$\boldsymbol{u} = \left(\frac{7.2}{0.41^4}yz(0.41-y)(0.41-z), 0, 0\right)^T,$$

at the outlet x = 2.55, the do-nothing boundary condition was imposed and on all other boundaries, the no-slip boundary condition was used. The flow field exhibits vortices behind the cylinder.

The Navier–Stokes equations were discretized with the popular Q_2/P_1^{disc} (continuous piecewise triquadratic velocity, discontinuous piecewise linear pressure) pair of finite element spaces on a hexahedral grid, see Figure 8 for the initial grid. Simulations were performed on level 2 (776 160 velocity d.o.f.s, 122 800 pressure d.o.f.s) and level 3 (6 052 800 velocity d.o.f.s, 983 040 pressure d.o.f.s). The nonlinear problem was linearized with a Picard iteration (fixed point iteration). In each step of this iteration, the iterative solvers reduced the Euclidean norm of the residual vector by at least the factor 10 before performing the next Picard iteration. The Picard iteration was stopped if the Euclidean norm of the residual vector was less than 10^{-8} .

In the geometric multigrid preconditioner, the so-called mesh-cell oriented Vanka smoother was used, e.g., see [23, 25]. This smoother is a block Gauss–Seidel method that solves a local system in each mesh cell. The multigrid preconditioner was applied with the V(2,2)-cycle.

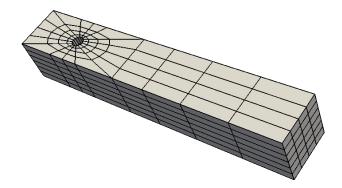


Figure 8: Example 3, initial grid (level 0).

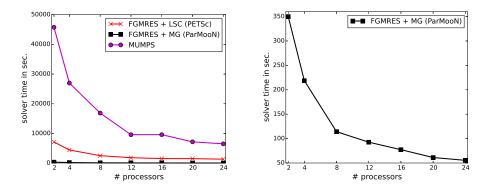


Figure 9: Example 3, solver times on refinement level 2.

The linearization and used discretization of the incompressible Navier-Stokes equations requires the solution of a linear saddle point problem in each Picard iteration. We tried several options for calling an iterative solver from PETSC for such problems, as well based on the coupled system as on Schur complement approaches. Only with a Schur complement approach and the least square commutator (LSC) preconditioner (-ksp_type fgmres -pc_type fieldsplit -pc_fieldsplit_type schur -pc_fieldsplit_schur_factorization_type upper -fieldsplit_1_pc_type lsc -fieldsplit_1_lsc_pc_type lu -fieldsplit_0_ksp_type preonly -fieldsplit_0_pc_type lu), reasonable computing times could be obtained, at least for the coarser grid, see Figure 9. Like for the scalar problems, the sparse direct solver performed by far the least efficient. The solver with the geometric multigrid method was more efficient by around a factor of 20 compared with the iterative solver called from PETSC. For the finer grid, the numerical studies were restricted to the geometric multigrid preconditioner, see Figure 10. It can be observed that increasing the number of processors from 2 to 20 reduces the computing time by a factor of around 6.6.

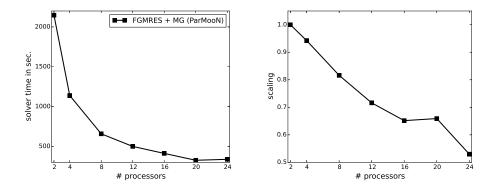


Figure 10: Example 3, solver times and scaling on refinement level 3. The scaling is computed by $2 \cdot t_2/(p \cdot t_p)$, where p is the number of processors and t_p the corresponding time from the left picture.

7. Summary

This paper presented some aspects of the remanufacturing of an existing research code, in particular with respect to its parallelization. All distinct features of the predecessor code could be incorporated in a straightforward way in the modernized code PARMOON. The efficiency of the most complex method in the parallel implementation, the geometric multigrid method, was assessed against some parallel solvers that are available in external libraries. The major conclusions of this assessment are twofold. For scalar convection-diffusion-reaction equations, the geometric multigrid preconditioner was similarly efficient as an iterative solver from the PETSC library. The larger the problems became, the better was its efficiency in comparison with the external solver. For linear saddle point problems, arising in the simulation of the incompressible Navier–Stokes equations, we could not find so far any external solver that proved to be nearly as efficient as the geometric multigrid preconditioner.

On the one hand, we keep on working at improving the efficiency of the geometric multigrid preconditioner. On the other hand, we continue to assess external libraries with respect to efficient solvers for linear saddle point problems, which can be used in situations where a multigrid hierarchy is not available. In addition, an assessment as presented in this paper on another widely available hardware platform, namely small clusters of processors, is planned.

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